IN THE SPECIFICATION:

Please amend the Specification as follows.

Please replace the second paragraph on beginning on page 10 and continuing to page 11 with the following new paragraph:

A turbine assembly 100 embodiment is described with reference to Figure 15. As indicated, the turbine assembly 100 includes a first component 110 and a second component 112. The first and second components 110, 112 define a ~~secondary~~ cooling slot 114. The ~~secondary~~ cooling slot 114 receives and guides a secondary coolant flow. Exemplary components 110, 112 that define a ~~secondary~~ cooling slot 114 include: a combustor and a turbine inlet nozzle, a combustor and a nozzle (stationary vane), a nozzle and a blade, a nozzle and a shroud, a blade and a shroud, two nozzles, and two blades. The turbine assembly further includes at least one flow modifier 30 formed on a surface of one of the first and second components 110, 112. For example, if the component is a blade, the flow modifier may be formed on the platform. If the component is a nozzle, the flow modifier may be formed on an end wall. If the component is a shroud, the flow modifier 30 may be formed on the shroud. The flow modifier 30 is adapted to enhance the secondary coolant flow along at least one of the first and second components 110, 112 within the ~~secondary~~ coolant slot 114. In this manner, the flow modifier 30 enhances the cooling of the components 110, 112 by the secondary coolant flow.

Please replace the second full paragraph on page 11 beginning at line 21 with the following new paragraph:

For the embodiment shown in Figure 15, the flow modifier 30 extends into ~~secondary~~ cooling slot 114. The flow modifier 30 is described above. According to a particular embodiment (not expressly shown), the flow modifier 30 forms a ridge 38 extending along one of the components 110, 112.